INTRODUCEDIN TO APPRIOUR D

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ITEM 1 - CIRCUIT BREAKER ITEM 2 - FUSE, AXIAL LEAD / CARTRIDGE ITEM 3 - FUSE, HIGH CHREAK ITEM 4 - FUSE, FLOG-IN

MC454-0026 / MC454-0032 NE451-0009 NE451-0016 HE451-0018

PATTABLE NOTES AND CAUGUES!

THE POLICHTING TRBLE LIGHTS PAILLING HOURS AND CAMBES WHICH WERE CONSIDERED IN DERIVING PAILING HOURS AND EFFECTS ANALYBLE (THEA'S) FOR THE ITEMS LIGHTED ABOVE;

PAULINS MODE	PAILUR CARE	CTRCUIT	PUER HC451-0009	FC188	PUER MC451-6018
FAILS OPEN, FAILS TO CHOUCE, FAILS TO CLOSE	(a) Structural Pailure (b) Contemination (c) Vibration (d) Mechanical Shock (e) Processing Amonaly (f) Thermal Stress	книни	 	жжжж	****
FALLS CLOSED, FALLS TO OPEN, (MESENAICALLY)	(a) Structural Fallure (b) Contamination (c) Vibration (d) Mechanical Shock (e) Processing Anomaly	ммжж	•	•	•
FAILS TO INTERBUPT UNDER OVERLOAD *	(a) Structural Failure	H			

* KESS DEETE PROSE.

APPENDIX D ITEM 1 - CIRCUIT BREAKER MC454-0026-XXXX, MC454-0032-XXXX

DISPOSITION & RATIONALE

- (A) DESIGN, (B) TEST, (C) INSPECTION, (D) FAILURE HISTORY:
- (A) DESIGN
 TRIP FREE, TEMPERATURE COMPENSATED, THERMAL CIRCUIT BREAKER.
 THIS CIRCUIT BREAKER IS DESIGNED, TESTED AND INSPECTED TO MEET
 THE REQUIREMENTS OF THE SPACE SHUTTLE PROGRAM DETAILED BY
 ROCKWELL INTERNATIONAL SPECIFICATION MC454-0031 AND THE
 REQUIREMENTS OF MILITARY SPECIFICATION MIL-C-5809 (MS3320). THE
 APPLICATION OF THE PART IS ANALYZED TO ASSURE COMPLIANCE WITH THE
 WIRE PROTECTION CRITERIA ESTABLISHED BY ROCKWELL ELECTRICAL
 DESIGN.

(B) TEST

QUALIFICATION/CERTIFICATION

CERTIFICATION TESTING AND ANALYSIS IS COMPLETED AND APPROVED. TESTS INCLUDE:

	CAUSE CONTROL							
TEST	a	Ъ	С	đ	e ·	f		
TRIP PERFORMANCE (110%, 145%, 200%)	Х				х	i		
VOLTAGE DROP	Х	х			x			
INSULATION RESISTANCE		Х			X			
DIELECTRIC STRENGTH (DWV, 1250 Vrms								
SINGLE PHASE, 1400 Vrms THREE PHASE)	٠.	X	ļ	l	Х			
ACTUATION FORCE (FULL OUT, FUSH IN)	X	•	1	}	Х			
ACTUATOR STRENGTH (25 LBS)	х		1		1			
VIBRATION (0.2 g ² /HZ, 3 AXES)	Х	1	X					
THERMAL (-65 TO 190 OF SOAK		1	-					
ALONG WITH 85 TO 190 OF)]				l	X		
CABIN ATMOSPHERE (HUMIDITY 80%, SALT			į	i	į	ļ		
FOG 1%, TEMPERATURE 60 TO 120 OF)		X	i	1		47.		
SAND AND DUST	t	X						
INTERRUPTING CAPACITY	Х	I			X			
SHOCK (40 G)		l		X	l			
OVERLOAD CYCLING (200%, 100 CYCLES)	l	1			X			
LIFE CYCLING (5000 CYCLES, MANUALLY)	Х	1			X			

APPENDIX D ITEM 1 CONT'D

ACCEPTANCE AND SCREENING

ACCEPTANCE TEST IS ACCOMPLISHED ON A 100% BASIS FOR ALL CIRCUIT BREAKERS. TESTS INCLUDE:

MDG/II	CAUSE CONTROL							
TEST		b	С	đ	e	f		
ACTUATION FORCE (PULL OUT, PUSH IN)	х				Х			
VIBRATION (0.04 g ² /HZ)	Х		х	1		ļ ,		
INSULATION RESISTANCE (IR AT 500 VDC) DIELECTRIC STRENGTH (DWV AT 1250 Vrms		X			X			
SINGLE PHASE, 1400 Vrms THREE PHASE)		X			Х			
TRIP PERFORMANCE (110%, 145%, 200%)	X			Į.	X			
VOLTAGE DROP	х	х		[Х			

ACCEPTANCE TEST AT THE NEXT ASSEMBLY LEVEL INCLUDE:

TEST		in the state of the		CAI	USE (CONT	ROL	
TEST	·		a	ф	. 0	d	ų	f
CONTINUITY ACTUATION (PULL OUT, PU	SH IN)		x	х			x	

(C) INSPECTION

RECEIVING INSPECTION (FAILURE CAUSE a,b)

UPON RECEIPT INSPECTION PERFORMS VISUAL AND DIMENSIONAL EXAMINATION OF ALL INCOMING PARTS AND VERIFIES THE MATERIALS USED. RECORDS ARE MAINTAINED CERTIFYING THE MATERIAL AND ITS PHYSICAL PROPERTIES.

CONTAMINATION CONTROL (FAILURE CAUSE b)

ULTPASONIC CLEANING OF DETAIL PARTS PRIOR TO FINAL ASSEMBLY AND A FILTERED CLEAN AIR BLAST ARE VERIFIED.

APPENDIX D ITEM 1 CONT'D

ASSEMBLY/INSTALLATION (FAILURE CAUSE a,b,e)

DETAILED INSPECTION IS PERFORMED ON ALL ASSEMBLIES PRIOR TO THE NEXT OPERATION. INSPECTION UNDER 4X MAGNIFICATION IS PERFORMED PRIOR TO CLOSEOUT. CIRCUIT BREAKERS ARE ASSEMBLED IN A DEDICATED AREA. DETAILED PARTS WITH DEFECTS ARE SCRAPPED WITH NO ATTEMPT TO REWORK.

CRITICAL PROCESSES (FAILURE CAUSE b,e)

MACHINE CONTROLLED SPOT WELDING OF CONTACTS TO TERMINALS AND BRAIDED WIRE TO BI-METAL. SET-UP INSPECTION AND PERIODIC IN-PROCESS INSPECTIONS PERFORMED.

TESTING (FAILURE CAUSE a,b,e,f)

ACCEPTANCE TEST PERFORMED BY CERTIFIED PERSONNEL, OBSERVED AND VERIFIED BY QUALITY CONTROL (QC) INCLUDING VIBRATION, FUNCTIONAL AND A MANUAL OPERATIONAL CHECK FOR BINDING.

HANDLING/PACKAGING (FAILURE CAUSE c,d)

USE OF CUSHIONED TRAYS VERIFIED FOR IN-PROCESS HANDLING. FARTS PACKAGED AND PROTECTED ARE VERIFIED BY INSPECTION TO APPLICABLE REQUIREMENTS.

(D) FAILURE HISTORY

SINGLE PHASE CONFIGURATION (MC454-0026-XXXX)

FAILURE MODE: FAILS OPEN, FAILS TO CONDUCT, FAILS TO CLOSE

CAR'S A8717, AB0031

FAILS TO CLOSE WHEN ACTUATED. CONTAMINATION, SILICON DIOXIDE, FOUND ON THE CONTACT SURFACES. THE SILICON DIOXIDE WAS FORMED BY A THERMAL REACTION FROM THE SILICON PRIMER USED TO COAT THE EXTERIOR SURFACES OF THE BREAKER. PRIMER ENTERED BREAKER BECAUSE IT WAS APPLIED IN AN INVERTED POSITION. PROCEDURES WERE REVISED TO APPLY PRIMER IN A NON-INVERTED POSITION.

CAR'S AB7060, AC4155, AC7395
WOULD CLOSE ELECTRICALLY BUT WOULD NOT MECHANICALLY LATCH.
ANALYSIS DETERMINED THE CAUSE TO BE A BROKEN MOVABLE CONTACT
SHORTING BAR. THE BAR WAS BROKEN DUE TO A MISALIGNMENT OF THE
CONTACT STRIP., CONTACT MATERIAL, SILVER-CADMIUM, SHEARED OFF AND
IMBEDDED INTO THE SOFTER BERYLLIUM COPPER CAUSING THE BERYLLIUM
COPPER TO BE THINNER IN THE AREA UNDER THE SHEARED OFF MATERIAL.
THE SUPPLIER REVISED QUALITY CONTROL TO INCLUDE A SAMPLE CHECK
AND VISUAL INSPECTION OF THE SHORTING BAR FOR CONTACT SHEARING,
THINNING AND PITTING.

APPENDIX D ITEM 1 CONT'D

CAR'S ABO766, AC1091, AC9246, AD0200
DURING ORBITER CHECKOUT TESTS SOME EQUIPMENT FAILED TO OPERATE OR
TURN "ON" DURING THE POWER-UP SEQUENCE. SUBSEQUENT
TROUBLESHOOTING FOUND THE CONDITION BEING CORRECTED WITH
MECHANICAL CYCLING OF THE ASSOCIATED CIRCUIT BREAKER. WHILE THE
FAILURES WERE NOT DUPLICATED, ANALYSIS REVEALED TRACES OF BRAZING
FLUX (BORO SILICATE) RESIDUE ON THE CONTACT SURFACES IN SOME
CASES. THIS RESIDUE COULD ACCOUNT FOR THE REPORTED ANOMALY,
ESPECIALLY SINCE IN THESE APPLICATIONS THE LOAD CURRENT IS VERY
LOW, LESS THAN 65 MILLIAMPERES.

TRAINING SESSIONS FOR INSPECTORS, ASSEMBLY AND SUPERVISOR PERSONNEL ARE CONDUCTED PRIOR TO PRODUCTION RUNS TO REINFORCE CLEANLINESS FRACTICES. IN ADDITION, ORBITER OPERATIONAL PROCEDURES NOW REQUIRE CYCLING OF THE CIRCUIT BREAKERS (FIVE TIMES) TO CLEAR HIGH CONTACT RESISTANCE CONDITIONS OF THIS NATURE.

THERE ARE NO UNRESOLVED GENERIC ISSUES.

THREE PHASES CONFIGURATION (MC454-0032-XXXX)

FAILURE MODE: FAILS OPEN, FAILS TO CONDUCT, FAILS TO CLOSE

CAR AB8733
BREAKER WILL NOT LATCH DOWN. CAUSED BY INTERNAL CONTAMINATION OF RTV PRIMER: THE PRIMER WAS APPLIED WITH THE BREAKER IN AN INVERTED POSITION WHICH IS IMPROPER. PROCEDURE HAS BEEN REVISED TO APPLY PRIMER IN AN NON-INVERTED POSITION.

CAR AC1737, AC22SO
FAILED TO LATCH MECHANICALLY WHEN ACTUATING BUTTON WAS PUSHED IN.
THE BREAKER WAS DAMAGED BY THE USED OF "CAPS" (DEVICE INSTALLED
ON BREAKER HANDLE) WHICH WERE INSTALLED BY TEST PERSONNEL.
CAUTION NOTE ADDED TO ALL KSC AND PAIMDALE PROCEDURES
RESTRICTING THE USE OF "CAPS".

CARS AC5128, AC5857, AC5951, AC7052, AC7053
BREAKER WILL NOT LATCH DOWN. CAUSED BY DAMAGE TO CLEVIS THAT WAS
CAUSED BY THE USE OF A TOOL TO OPEN/CLOSE BREAKERS. PROCEDURES
HAVE BEEN CHANGED TO PROHIBIT THE USE OF TOOLS ON CIRCUIT
BREAKERS.

CAR 02F019
DURING STS-2 OSTA PALLET PUMP PRESSURE WAS NOT HIGH ENOUGH.
COULD NOT VERIFY FAILURE. PROBLEM COULD HAVE BEEN CAUSED BY
RESIDUE ON CIRCUIT BREAKER CONTACTS. THE CREW WILL BE ADVISED TO
TRY CYCLING BREAKER IF A SIMILAR PROBLEM SHOULD OCCUR.

THERE ARE NO UNRESOLVED GENERIC ISSUES.

11/20/87 (11:04am)

APPENDIX D ITEM 1 CONT'D

PREPARED BY:

AFFROVED BY:

APPROVED BY (NASA):

DESIGN

D. BUENDIA M. HOVE

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RELIABILITY J. COURSEN QUALITY

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